

TOROUES IMAIRCRAFT DEVELOPMENT

AIRPLANE INFORMATION MANUAL

NOT FOR FLIGHT - FOR SIMULATION USE ONLY

Revision C 11 June 2020





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TS-BN-2-AIM Intro-II

Acknowledgments

Special Thank You

Valentin for amazing sound recordings

Saso Kiselkov for libacfutils (https://github.com/skiselkov/libacfutils), and for help all along the development process!

Our Customers: Thank you for supporting us – We look forward to integrating your feedback and making this plane even better!

Our Testing Team

Andrew H. Johnny K. Rafael S.

Christoph T. Matteo P. Steven M.

Cullen P. Matthew C. Valentin D.

Grau A. Moritz R.

Developers

Ulrich Oetzel 3D Model, Textures, Liveries, Flight Model, and more

Cooper LeComp Systems Development, Avionics, Miscellaneous Tuning

Steaven McKenzie FMOD Sounds, 3D People, Art Consultation



The BN-2 Islander is Proudly Sold Exclusively through X-Aviation

TS-BN-2-AIM Intro-III

Section 1 General

Table of Contents

Introduction	2
The Airplane	3
Engine	3
Propeller	3
Fuel	3
Oil	3
Performance	4

Introduction

The BN-2 Islander is a light twin designed in the 1960's as a regional airliner and utility aircraft, and is still widely flown today. Thanks to its rugged construction and excellent short field performance, this aircraft can be found with operators all over the world. The model simulated is the BN-2B-26, equipped with 2 Lycoming O-540 engines. The plane has an 1800 pound useful load with full tanks and can carry 9 passengers on missions with the shortest of runways. Additionally, a 730 nautical mile maximum range and 140 knot cruising speed give the Islander enormous operational flexibility. This airplane is the perfect addition to any simmer's hangar!

The TorqueSim rendition of this fantastic aircraft includes many simulated features designed to enhance your simulation experience. In addition to the beautiful 3D and textures, there are many custom systems including a fully custom electrical system, fuel manager, lighting, and more! Avionics were picked to represent a true workhorse aircraft. The KFC225 Autopilot, GTX345 Transponder, KR87 ADF, and M800 Chronometer were build custom for the aircraft. The default GNS530 and GNS430 is also installed as the GPS of choice.

Customer Support Information

X-Aviation (Sales, Installation, Activation): torques.im/xasupport

X-Pilot Forum (Support, Discussion, Help): torques.im/forum

TorqueSim (Aircraft Problems, Comments): torques.im/support

Bug Reporter (Discovered Issues): torques.im/bugs

The Airplane

Engine

Number of Engines	2
Number of Cylinders	6
Engine Manufacturer	Lycoming
Engine Model	O-540-E4C5
Fuel Metering	Carburetor
Horsepower Rating	260 hp @ 2700 rpm
Bore, Stroke (Inches)	5.125, 4.375
Displacement (Inches)	541.5
Compression Ratio	8.5:1
Fuel Consumption (59% Power, USG/hr)	11.4
Propeller	
Propeller Manufacturer	Hartzell
Propeller Type	Constant Speed, Two Blade
Model Number	HC-C2YK-2B/C8477-4
Feathering	Fully Feathering
Fuel	
Total Capacity	138 U.S. Gallons
Approved Fuel Grades:	
100 LL Grade Aviation Fuel (Blue)	
100 (Form. 100/130) Grade Aviation Fuel (Green)	
Oil	
Oil Capacity (Sump)	12 U.S. Quarts
Oil Grades:	

 Below 40°F (4°C)
 SAE 30

 Above 40°F (4°C)
 SAE 50

Performance

(At 6,300 lb. Take-Off Weight)

Take-off Run (ft)	560
Take-off Distance over 50 ft. obstacle (ft)	1,090
Minimum Control Speed - Single engine (IAS kts)	40
Best Rate of Climb Speed (IAS kts)	65
Rate of Climb (fpm)	1,050
Best Angle of Climb Speed (IAS kts)	65
Best Single Engine Rate of Climb Speed (IAS kts)	65
Single Engine Rate of Climb (left engine feathered) (fpm)	190
Absolute Ceiling (ft)	16,200
Service Ceiling (ft)	14,600
Single Engine Absolute Ceiling (left engine feathered) (ft)	5,600
Single Engine Service Ceiling (left engine feathered) (ft)	2,500
Top Speed (IAS kts)	183
Optimum Cruising Speed (IAS kts)	140
Stalling Speed - Flaps down (IAS kts)	38 to 43
Landing Roll (ft)	450
Landing Distance over 50 ft. Barrier (ft)	960
Cruising Range (maximum at 59% power at 13 000 ft) (miles)	840

Section 2 Design Information

Table of Contents

Electrical System2	Avionics7
Battery 2	M800 Chronometer 7
Alternators 2	KFC225 Autopilot7
Emergency Bus2	KR87 ADF7
Main Bus 3	GTX345 Transponder8
Avionics Bus 3	Cockpit Features8
Starter4	HDR Lights8
Main Switch Panel 4	Overhead Utility Lights 8
Circuit Breaker Panel4	Avitab Integration9
Engines5	InHg/mmHg Switch9
Airframe Systems5	Automatic Laminar 530/430 to RXP
Deicing Systems5	GTN 750/650 Change
Fuel System 6	Noise Cancelling Headphones 10
Dual Vacuum System 6	

Electrical System

The electrical system on the BN-2B-26 Islander consists of 1 lead-acid battery, and 2 alternators, one on each engine. The aircraft is equipped with three busses, the Emergency Bus, the Main Bus, and the Avionics Bus.

Battery

The Islander is equipped with one 25-amp-hour lead-acid battery. This battery is capable of powering the plane for short durations before engine start-up and after engine shut-down. During phases of flight when the alternators are running, they should be engaged and the battery will charge. In the case of dual-alternator failure, the battery is capable of powering essential systems for a duration based on electrical load, landing as soon as possible and reducing electrical usage to the absolute minimum is required.

Alternators

The Islander is equipped with two alternators, one on each engine. Both alternators are capable of providing 50 amps of current to the aircraft. The electrical load will naturally balance between both alternators when both are running.

Emergency Bus

The Islander's main electrical bus is the Emergency Bus. The top row of circuit breakers are tied to the electrical bus.

The electrical items on this bus are:

- Auxiliary Fuel Pump 2 (Port and Starboard)
- Fuel Contents Indicators (Port and Starboard)
- Nav Lights
- Map Light
- · Turn and Bank Indicator
- Stall Warning
- · Stall Heater
- · Pitot Heater
- Tachometer
- Audio Panel
- VHF 1 Radio (Part of GNS 530)

Main Bus

The Islander's secondary bus is the Main Bus. The circuit breakers on this bus are on the left and middle of the second and third row, and the whole of the auxiliary panel. The bus is interconnected through the Main to Emergency Busbar Isolation.

The electrical items on this bus are:

- Auxiliary Fuel Pump 1 (Port and Starboard)
- Landing Light (Port and Starboard)
- Cabin Lights
- · Panel Lights
- Beacon Lights
- Wing Flaps
- Tank Selector (Left and Right)
- Starter Relays
- Engine Cluster (Port and Starboard)
- Airframe deice
- · Cabin heat
- Rudder Servo
- · Aileron Servo
- Elevator Servo
- · Electric Trim

Avionics Bus

The Islander's tertiary bus is the Avionics Bus which powers the main avionics onboard. The Avionics Bus is connected to the Main Bus through the Avionics switch. The Avionics switch should be turned off during the connection and disconnection of alternators and when starting the engines.

The electrical items on this bus are:

- ADF
- VHF 2 Radio (Part of GNS 430)
- Marker Beacon
- Autopilot (KFC225)
- Transponder (GTX345)
- RMI
- GPS 1 (GNS 530)
- GPS 2 (GNS 430)
- DME

Starter

The starter is connected to the battery through the External Power/Starter Isolation switch. This switch must be turned on for the starters to function. This is a safety feature to prevent inadvertent starter engagement.

Main Switch Panel



Circuit Breaker Panel



Engines

The Islander uses two Lycoming O-540-E4C5 engines putting out 260hp each. Each engine is equipped with dual magnetos, a generator, and engine-driven fuel pumps and vacuum pumps.

The Islander is capable of flight on a single engine, though maximum altitude and climb performance are severely reduced, and this mode should not be used in any case except emergencies.

Airframe Systems

Deicing Systems

This aircraft can optionally be equipped with anti-ice boots for the wings, tailplane, rudder, and propeller leading edges for flight in cold climates. The Islander is not certified for flight into known icing, however, and no such maneuver should be attempted.

The deicing systems can be turned on or off from within the UI Menu system accessible from the left edge of the screen. While they are off, all deice boots, switches, and associated systems are hidden from view and cannot be activated.

Pitot and stall warning heating systems come standard with the aircraft.





Deicing switches when equipped and hidden

Fuel System

The wings each house a single integral fuel tank able to hold 65 gallons of Avgas usable, for a total of 130 gallons. Dual fuel selectors overhead the cockpit allow for each engine to feed from either tank independently.

Under normal flight conditions, crossfeed is not required, and the tanks should drain equally throughout a flight. It is not recommended to apply single-engine fuel tank selection procedures during flight (ie changing tanks every 20 minutes).

During startup, takeoff, and and landing, the auxiliary fuel pumps should be operated to support and back up the engine-driven fuel pumps. Auxiliary fuel pumps in this model will boost fuel pressure to around 6 psi.



Fuel Selectors and Fuel Pump Switches



Fuel Pressure Indicators

Dual Vacuum System

The vacuum system of the Islander is cross-tied. Should one pump fail, the second one will maintain enough suction in the system to allow normal operation of all dependent instruments.

Avionics

M800 Chronometer

The M800 Chronometer has all three modes (UTC, Local Time, and Elapsed Time) are simulated. The mode "UT" will show the Simulator's UTC time. The mode "LT" will show the Simulator's Local time. The mode "ET" will show the Elapsed time, both count down and count up. To switch between modes, click the "Select" button. In the ET mode, clicking the "Control" button will start a count up timer. Clicking again will stop the timer, and clicking once more will reset the timer. Clicking and holding the button for three seconds will enter count down set mode. The "Control" button will adjust the digit, then "Select" will cycle through the four digits before starting the timer.







M800 Chronometer in UT mode, LT mode, and ET mode

KFC225 Autopilot

The KFC225 Autopilot includes a Yaw Damper, Heading Hold, Nav mode, Approach mode, Reverse Course mode, Altitude Hold mode, and Vertical Speed mode. The autopilot is equipped with an altitude preselector. Rotating the "Alt Sel" knob will adjust the altitude, the outer knob in 1000 ft increments, the inner knob in 100 ft increments.

To arm the altitude selected, you **MUST** click the "ARM" button, otherwise the autopilot will not capture the altitude selected.

KR87 ADF

The KR87 Mode includes the primary ADF selection and standby ADF selection. Rotating the right knob will adjust the standby selection, the outer knob incrementing the selection in 100 increments, the inner knob in 1 increments. Clicking the FRQ/<-> Button will switch the standby frequency into the primary and the primary frequency into the standby.

The GTX345 Transponder is equipped with 3 main modes, Standby (SBY), On, and Altitude Reporting (ALT). Clicking these three buttons will set the transponder mode. The IDNT button will Ident the transponder. The VFR button will set the transponder to 1200. Clicking the button a second time will switch to 2000.

Pressure altitude is displayed to the right of the transponder code selection.



KFC225, KR87, and GTX345

Cockpit Features

The 1.1 update of the Islander has a few new cockpit features, which are outlined here.

HDR Lights

The Islander has HDR lights

Overhead Utility Lights

The overhead utility lights for both the pilot and copilot are functional. A small switch on the back of each activates the light itself, and the beam can be directed almost anywhere in the cockpit by clicking on and dragging the body of the light itself.

By nature of their function, these are HDR lights, which means that effects settings in X-Plane must be 'High' or 'Maximum'. On 'Medium' or below, these lights will not work! This is a limitation of X-Plane's rendering system.

Avitab Integration

The excellent Avitab tablet has been integrated with the 3d cockpit. Initially, it starts out hidden from view on every flight, but can be brought up by clicking on the glare-shield edge above the engine gauges.

By clicking and dragging the top and left edges of the tablet, it can be moved to a variety of positions above the glare-shield.

InHg/mmHg Switch

A click on the face of the pilot side altimeter switches the baro display between inches of mercury and millimeters of mercury.

Automatic Laminar 530/430 to RXP GTN 750/650 Change

This function requires a correct installation of the RXP GTN 750 and/or GTN 650 to work. If the automatic switch does not happen, please ensure that your RXP products are installed correctly. We cannot give assistance for any incorrect installation.

If you own (and have correctly installed) the RXP GTN 750 or GTN 650, you can use these in the Islander 3d cockpit. Simply select them from the Plugins menu, and assign any of the 7xx to GTN1 and any of the 6xx to GTN2. The RXP GTNs are integrated with both the normal and G5 versions of the Islander.

Be aware that limitations in the 3d model mean that you cannot use the GTN 6xx in the GTN1 slot or the GTN 7xx in the GTN2 slot.



Noise Cancelling Headphones

The headphones in the Islander model feature adjustable noise cancellation. With the headphones active, simply click and drag the highlighted manipulator or scroll with the mouse wheel to adjust the level of cancellation.

This setting is saved between sessions.



Section 3 Normal Procedures

Table of Contents

Preflight Checks 2
Walkaround2
Cabin2
Left Engine2
Left Undercarriage2
Left Leading Edge2
Left Trailing Edge3
Fuselage Left Side3
Tail Surfaces3
Fuselage Right Side3
Right Trailing Edge3
Right Leading Edge3
Right Undercarriage3
Right Engine3
Fuselage Nose3
Pre Starting Checks 4
Starting5
After Starting 6

Taxi	.6
Pre-Take-Off Checks	.7
Take-Off Procedure	.7
Handling In Flight	.8
Use of Carburetor Heat	8
Instrument Flying	8
Flying in Low Temperature Conditions	. 8
Flight at Reduced Speed	8
Flying in Turbulence	8
Stalling	8
Pre Landing Checks	.9
Final Approach	.9
Balked Landing	.9
After Landing Checks	10
Engine Shutdown	10
After-Shutdown Checks	10

Preflight Checks

Walkaround

Before starting the walk around inspection remove the pitot head cover, control locks, and battery vent plug. During the external inspection, perform a general check of airframe condition and the security of all fasteners and control hinges and attachments.

Cabin

Magneto Switches	ALL OFF
Parking Brake	ON
All Electrical Switches	OFF
Battery Master Switch	ON
Pitot Heater Switch	ON
Stall Warning Vane	Deflect and listen for stall horn.
Pitot Head	Check heat.
Rotating Beacon	ON, CHECK, OFF
Navigation Lamps	ON, CHECK, OFF
Battery Master Switch	OFF

Leaving the Battery Master Switch in the on position for an extended period of time during walkaround can drain the battery. If this happens, use 'Charge Battery' in the UI Menu to restore power.

Left Engine

l off lindoroovrions	
Gascolator	DRAIN AS NECESSARY
Oil	CHECK
Propeller	CONDITION

Left Undercarriage

Oleo	EXTENSION
Brake Discs and Pads	CONDITION
Brake Lines	CHECK FOR LEAKAGE
Tires	INFLATION, CUTS, CREEP

Left Leading Edge

Pitot Head	PRESSURE, STATIO	VENTS, DRAIN HOLE CLEAR
De-Icer Boots		CONDITION

Left Trailing Edge

Aileron	FREEDOM
Fuel Sump	DRAIN AS NECESSARY
Fuselage Left Side	
Passenger Doors	SECURED, UNLOCKED
Baggage Bay Door	SECURED, UNLOCKED
Tail Surfaces	
Rudder/Elevator	
Trim Tabs	CONDITION
Fuselage Right Side	
Passenger Door	SECURED, UNLOCKED
Right Trailing Edge	
Fuel Sump	DRAIN AS NECESSARY
Aileron	FREEDOM
Right Leading Edge	
De-Icer Boots	CONDITION
Right Undercarriage	
Oleo	EXTENSION
Brake Discs and Pads	CONDITION
Brake Lines	CHECK FOR LEAKAGE
Tires	INFLATION, CUTS, CREEP
Right Engine	
Propeller	CONDITION
Oil	CHECK
Gascolator	DRAIN AS NECESSARY
Fuselage Nose	
Nose Oleo	EXTENSION
Nose Wheel	INFLATION, CUTS, CREEP
Nose Cone	CONDITION
Windscreen	CLEANLINESS, CONDITION

Pre Starting Checks

Battery/Master Switch	ON
Pilot's Seat	ADJUST, SECURE
Rudder Pedals	ADJUST, SECURE
Doors	CLOSED AND LOCKED
Safety Belts	FASTENED
Parking Brake	ON
Trim Tabs	FULL MOVEMENT, NEUTRAL
Fuel Selectors	FREEDOM, ON
OAT	CHECK
Altimeters	CROSS CHECK
Timepiece	CHECK AND SET
Cabin Heater	AS REQUIRED
Fuel Contents Indicators	CHECK
Compasses	CHECK
Circuit Breakers	PUSHED IN
Flight Controls	FREEDOM
FlapsOperate over full range	e. Check visually and against indicator.
Avionics	OFF
External Lights	AS REQUIRED
Throttles	CLOSED
Propeller Controls	FULL FORWARD
Mixture Controls	FULL RICH
Carburetor Heat	FREEDOM, OFF
External Supply Switch	EXTERNAL SUPPLY (UP)
Aux Fuel Pumps	OFF
Generators	OFF

When starting from internal battery, the External Supply switch must be turned on to permit the battery to power the starters. Once start has been achieved, return the switch to the OFF/STARTER ISOL position.

WARNING

AVOID HEAVY PROLONGED USE OF THE STARTER. IF THE ENGINE IS RELUCTANT TO START AFTER ABOUT SIX REVOLUTIONS, THEN REST THE SYSTEM TO PREVENT OVERHEATING OF THE COMPONENTS.

To prevent overheating the air cooled Lycoming engine on the ground, the following precautions are strongly advised.

- Avoid unnecessary ground running.
- 2. Ensure the airplane faces the wind.
- 3. Confine propeller pitch to the full fine position.
- 4. Maintain at least 1200 RPM to avoid fouling the spark plugs.
- 5. Keep the mixture control in the full rich position.

Starting

The right hand engine is normally the first to be started. Select the starter isolation switch to ON (UP) prior to using the starter.

Engine should fire almost immediately on application of the starter.

Right Magneto ON

Throttle 1000-1200 RPM

Oil Pressure Indicating within 30 Seconds

Repeat for left engine.

After Starting

Aux Fuel Pumps	OFF
Generators	ON, Check for low-volt lights extinguished
Vacuum Gauge	3-5 in Hg
Ammeter	Indicator within green sector
Ammeter Selector	BATT
Oil Pressure	ABOVE RED LINE MINIMUM
Fuel Pressure	WITHIN GREEN SECTOR
Flight Instruments	Indicating normally
Avionics	ON
Cabin Heat	AS REQUIRED
Ice Protection Systems	CYCLE, OFF

After engines are warmed up

- Test for magneto dead cut at 1200 RPM.
- 2. Test magneto drop off at 17 in manifold pressure (2100 RPM). Drop should not exceed 175 RPM.
- 3. Exercise propeller pitch controls at 2100 RPM.
- 4. Feathering check at 1500 RPM. Return prop control lever to normal range before RPM decreases below 1000.
- 5. Check ammeter reading 2-5 amps in the green sector. Check generator balance.
- 6. Check voltmeter for voltage between 27 and 29 volts.

Taxi

Test brakes for effectiveness during taxi. At the runway threshold, set engine power to ~70% and select carburetor head FULL; check for manifold pressure drop and return the heat control to OFF (UP).

Pre-Take-Off Checks

Passenger Notices	ON
Doors and Windows	CLOSED
Propeller Control	FULL FINE
Carburetor Heat	OFF
Trims	NEUTRAL
Flaps	T.O.
Fuel	Check contents, selection, and pressure.
Aux Fuel Pumps	ON
Aux Fuel Pumps Engine Instruments	
	CHECKED
Engine Instruments	CHECKEDCHECKED AND SET
Engine Instruments Flight Instruments	CHECKEDCHECKED AND SETON AS REQUIRED
Engine Instruments Flight Instruments Pitot Heater	

Take-Off Procedure

Advance throttles slowly to the open position. Rotate the aircraft at a speed 4 knots less than the scheduled 50 ft safety speed. The aircraft is capable of taking off in a 30 knot crosswind without resort to differential throttle.

Climb out should occur at no less than 64 knots. Avoid retracting the flaps at a height of less than 200 ft AGL, as the aircraft may experience a brief sink during flap retraction. Trim for resultant nose heaviness.

Turn Aux fuel pumps off at a safe altitude.

Engine RPM should be reduced to 2500 RPM to reduce noise and vibration. Throttle may be maintained in the full open position during climb. Maximum climb power is at 2700 RPM.

Lean mixture as necessary to maintain good power and fuel economy.

Handling In Flight

The airplane is easy to fly at all speeds and has no unusual features.

Use of Carburetor Heat

Carburetor icing can occur unexpectedly in various combinations of atmospheric conditions. In damp, foggy, or cloudy conditions, keep a sharp observation for power loss indicative of carburetor icing. Should icing arise, apply full carburetor heat for 30 second, then return the controls to OFF (UP) to avoid excessive power loss.

Partial application of carburetor heat should only be done to prevent carburetor ice from forming. Should icing be suspected, FULL heat should be applied first to clear it.

Instrument Flying

The aircraft is equipped for IFR flight.

Flying in Low Temperature Conditions

Propeller governors should be exercised regularly to keep the oil supply circulating and thus avoid sluggish or unresponsive propellers.

Flight at Reduced Speed

A comfortable attitude can be maintained at 78 knots by setting RPM at 2400 and lowering flaps to T.O. position. Airspeed in this configuration must not exceed 114 knots IAS.

Flying in Turbulence

The recommended air speed during turbulence is 88 knots IAS.

Stalling

Stalls are gentle in all configurations and from all attitudes normally encountered. Recovery is affected by pushing the nose down. Provided action is taken promptly, altitude loss will be minimal.

The stall horn and warning annunciator will activate at a safe margin above stall.

Pre Landing Checks

Harness	FASTENED
Pilot's Seat	SECURE
Fuel	CONTENTS, SELECTION
Engine Instruments	CHECKED
Mixture	FULL RICH
Carburetor Heat	OFF or INTERMITTENT
Propellers	FULLY FINE
Aux Fuel Pumps	ON, CHECK PRESSURE
Warning Lamps	EXTINGUISHED
Autopilot	OFF
Passenger Notices	ON
Brakes	OFF
Pitot Heat	ON AS REQUIRED

Final Approach

After selection of flaps down (56 degrees), the speed may be progressively reduced to the appropriate threshold speed.

Approach can be flown comfortably at 65 knots at all permissible weights and CG conditions. This will result in a slight nose high attitude with very little flare required prior to touchdown.

Balked Landing

Apply full power smoothly and establish a positive rate of climb. Set flaps to T.O. and accelerate above 61 knots. Flaps should not be retracted before 200 ft AGL and 65 knots.

WARNING

FULL FLAP DEFLECTION CAUSES VERY HIGH DRAG AND MAY REDUCE CLIMB PERFORMANCE. IMMEDIATE FULL RETRACTION AT LOW SPEED MAY LEAD TO FURTHER ALTITUDE LOSS.

WARNING

APPLICATION OF FULL POWER WITH FULL FLAP DEFLECTION CAUSES A SIGNIFICANT NOSE UP MOMENT.

After Landing Checks

Aux Fuel Pumps	OFF
Flaps	UF
Pitot Heat	OFF
Unrequired Services	OFF
Trim	NEUTRAL
Cabin Heater	OFF

Engine Shutdown

Idle at 1000 RPM until temperatures drop and check the magnetos for a dead cut. Stop the engines by moving the mixture controls to the cutoff position.

WARNING

FAILURE TO TURN OFF THE AVIONICS PRIOR TO STOPPING THE ENGINES CAN RESULT IN DAMAGE TO THE ELECTRONICS.

After-Shutdown Checks

Fuel Selectors	OFF
Magnetos	OFF
Generators	OFF
Other Services	OFF
Battery Master Switch	OFF
Parking Brake	ON
Chocks	IN POSITION

Section 4 Emergency Procedures

Table of Contents

Electrical Failures 2	Handling on One
Failure of Right Generator2	General
Failure of Left Generator2	Fuel System
Failure of Both Generators2	Unfeathering
Failure of Main Busbar3	Critical Engin
Failure of Battery3	Landing with
Fires	Emergency Evac
Fire on the Ground - Engine Fire4	Operation of
Fire in the Air - Fire in the Cabin4	Operation of
Fire in the Air - Engine Fire4	
Engine Failure5	
Failure of One Engine during Take Off 5	
Failure of One Engine after Take Off5	
Immediate Action	

Ha	andling on One Engine	. 5
	General	. 5
	Fuel System Management	. 6
	Unfeathering Sequence	. 6
	Critical Engine	. 6
	Landing with One Engine Inoperative	. 6
En	nergency Evacuation of Aircraft	. 7
	Operation of Cabin Doors	. 7
	Operation of Cabin Window Exits	. 7

Electrical Failures

Failure of Right Generator

Right Generator Field SwitchOFF
Right Generator Circuit BreakerTRIP

Failure of Left Generator

No load shedding is necessary except at the pilot's discretion.

Failure of Both Generators

OFF
OFF
TRIP
TRIP
ON

Check that the circuit breakers for the following services are engaged and that the services are operating or ready for use as indicated:

- a. Auxiliary fuel pumps Limit to 5 minutes for landing only
- b. Wing flaps Limit to essential operation only
- c. Fuel contents indicators
- d. Landing lights Limit to 2 minutes for landing only
- e. Navigation lights Night operations only
- f. Engine instruments cluster
- g. Map light Night operations only
- h. Turn-and-bank indicator
- i. Stall warning
- j. Rotating beacons
- k. Stall warning heater
- Pitot head heater
- m. Magnesyn compass
- n. Tachometer
- o. Audio system
- p. One essential radio navigational aid
- q. One essential radio communications transmitter/receiver Limit to 5

minutes use during cruise and 2.5 minutes minutes while landing

All other electrical services......OFF/CIRCUIT BREAKERS TRIPPED

Under the above conditions the aircraft battery should have sufficient capacity to allow a flight of 25 minutes duration at night or 30 minutes duration by day.

Failure of Main Busbar

Left Generator Field Switch	OFF
Right Generator Field Switch	OFF
Left Generator Circuit Breaker	TRIP
Right Generator Circuit Breaker	TRIP

Failure of Battery

Battery Switch	OFF
Left Generator	ON
Right Generator	ON
Pitot and Stall Heaters	OFF
Nav Lights	OFF
Landing Lights	OFF
Cabin Lights	OFF
Passenger Notices	OFF
Rotating Beacons	OFF
All De-Ice systems	OFF
Cabin Heater	OFF

When electrical loading of the main busbar is reduced sufficiently, both left and right generators will come back online; this can be verified by the ammeter pointer registering in the green sector when selected to the generators, and the voltmeter registering between 27 and 29 Volts. As the generators are not completely stable without the battery, heavy loads may cause them to go offline again, at which point electrical load should be reduced until generator power is restored.

Fires

Fire on the Ground - Engine Fire

Mixtures	OFF
Magnetos	OFF
Battery Master	OFF
Fuel Tanks	OFF

Deal with the fire by using the fire extinguisher bottle located beneath the pilot's seat or any similar ground appliance.

Fire in the Air - Fire in the Cabin

Deal with the fire immediately using the fire extinguisher located under the pilot's seat. Although the fumes given off by the extinguisher are non toxic, the cabin should be well ventilated by opening the direct view window and selecting the cabin ventilating blower ON as soon as the fire has been completely extinguished.

Fire in the Air - Engine Fire

On the first indication of a fire, take the following steps after identifying the affected engine:

Mixture	CUT OFF
Propeller	FEATHER
Throttle	CLOSED
Fuel Tank	OFF
Magnetos	OFF
Fuel Pumps	OFF
Generator	OFF

When the fire has died away, do not attempt to restart the affected engine. In the case of a persistent fire, a landing must be made at the first available opportunity.

Engine Failure

Failure of One Engine during Take Off

If one engine fails before the take off safety speed is reached, close the throttles and brake to a stop.

Failure of One Engine after Take Off

Immediate Action

In the event of an engine failing after take off safety speed is reached, and whilst the aircraft is climbing, the following procedure must be effected:

Identify the inoperative engine

Mixture	OFF
Propeller	FEATHER
Throttle	CLOSED
Fuel Tank	OFF
Magnetos	OFF
Fuel Pumps	OFF
Generator	OFF

Ensure that the generator of the operative engine is ON. Allow airspeed to build to 65 kt prior to selecting flaps UP.

Warning

It is essential to raise the flaps to the fully up position to achieve the optimum climb gradient

Handling on One Engine

General

The aircraft is perfectly docile on one engine and should maintain a height of 5600 ft at a gross weight of 6300 lbs in ISA conditions.

Fuel System Management

Should an engine failure be experienced during the early part of the flight the operative engine can be fed, alternately, from either fuel tank to balance the fuel load. This can be done by using the tank selector, applicable to the operating engine, as indicated by the inscribed markings. It is recommended that the tanks be switched at intervals of approximately 30 minutes.

Unfeathering Sequence

Identify the engine to unfeather.

Fuel Tank	ON
Magnetos	ON
Fuel Pumps	ON
Throttle Lever	
Propeller	CRUISE SETTING
Prime engine2-4	PUMPS OF THE THROTTLE
Isolation Switch	ON/UP
Mixture	RICH
Starter	ACTIVATE
When engine fires, select right magneto swit	
When engine mos, coloct fight magnete owit	ch ON.
Check engine TEMPS and PRESSURES.	ch ON.
Check engine TEMPS and PRESSURES.	JUST TO CRUISE SETTINGS
Check engine TEMPS and PRESSURES. PropellerAD	JUST TO CRUISE SETTINGS

Critical Engine

Failure of the left engine has the most adverse effect on the handling and performance characteristics of the engine.

Landing with One Engine Inoperative

Make an initial approach at 65 kt IAS with the flaps selected to T.O.

When committed for landing, select FLAPS DOWN and reduce speed

Emergency Evacuation of Aircraft

Operation of Cabin Doors

All three cabin doors are placarded as emergency exits and should be opened in the normal way as the first resort in making an exit from the cabin. In the case of the pilot's door, however, the left engine magnetos should first be switched OFF, thus allowing the solenoid operated interlocking mechanism to be released. Should this action be overlooked, or if there is an insufficient time available, the interlocking mechanism can be overridden by operating the lock-release lever with one hand whilst operating the internal door handle with the other hand.

Operation of Cabin Window Exits

If the cabin doors cannot be opened, and particularly if the aircraft comes down on to water, each emergency 'pull-in' window must be removed as follows:

- Grip the red handle at the top corner of the cabin door window trimming
- 2. Pull inwards and rearwards as hard as possible.

It is necessary to dislodge the window from its groove in the rubberized moulding and considerable physical effort may be required, especially in cold conditions.

Section 5 Performance Data

Table of Contents

Takeoff Performance	2
Takeoff Safety Speeds	
Takeoff Distance	2
Maximum Takeoff and Landing Weight	3
Cruise Performance	
Landing Performance	5
Landing Distance	5
Safe Landing Speed	5

Takeoff Performance

Takeoff Safety Speeds

TAKEOFF SAFE	TY SPEED
WEIGHT	KNOTS
4000	48
4500	50
5000	51
5500	53
6000	54
6500	56
7000	57

Takeoff Distance

TAKEOFF DISTA	ANCE IN FT AT ISA	+30 C		
	TOW			
ALTITUDE	4000	5000	6000	6600
0	850	1150	1450	1600
1000	1000	1300	1600	1750
2000	1150	1450	1700	1900
3000	1250	1550	1850	2000
4000	1400	1700	1950	2150
5000	1500	1800	2050	2250
6000	1600	1950	2200	2400
7000	1750	2100	2350	2500

^{*}SUBTRACT 50FT FOR EVERY 10 C LOWER
WIND AND AIRFIELD SLOPE ARE NOT CONSIDERED IN THIS CHART

Maximum Takeoff and Landing Weight

MAXIMUM TAKEOFF AND L	OFF AND LA	ANDING WEI	GHT (LBS) E	Y ALTITUDE	ANDING WEIGHT (LBS) BY ALTITUDE AND TEMPERATURE	ERATURE	
	TEMPERATURE	JRE					
ALTITUDE FT	ISA -10 C	-2 C	0 0	10 C	20 C	30 C	ISA +30 C
0	0099	0099	0099	0099	0099	0089	0009
1000	0099	0099	0099	0059	6250	6050	2120
2000	6450	6450	6450	6250	0009	2800	2220
3000	6250	6250	6250	0009	2750	5550	5350
4000	0909	6050	2950	5750	2200	5300	5150
2000	2850	5850	5750	2200	5300	5100	4950
0009	2650	2600	2200	5250	2050	4850	4800
2000	5450	5350	5250	5050	4850	4650	4600

TEMPERATURES ARE ACTUAL AIR TEMPERATURE UNLESS OTHERWISE INDICATED

Cruise Performance

		STANDAE	RD DAY C	STANDARD DAY CONDITIONS	S									
		MANIFOL	MANIFOLD PRESSU	URE - INC	IRE - INCHES OF MERCURY	MERCUR	>							
		2700	2600	2500	2400	2300	2200	2100	2000				FUEL CONSUMPTION GAL/HR	TION GAL/HR
POWER	DENSITY ALTITUDE	RPM	RPM	RPM	RPM	RPM	RPM	RPM	RPM	AS MPH	RPM IAS MPH TAS MPH TAS KTS	TAS KTS	US RICH	US LEAN
75 PERCENT	8000	20.50								146	161	140	23.5	29.0
195 BHP	0009	21.25	21.75	22.25	23.00					149	159	138	33.5	29.0
	4000	21.75	22.25	22.75	23.25	24.25				151	157	137	35.0	NOT <5000 FT
	2000	22.25	22.75	23.25	24.00	24.75	25.75			153	154	134	36.0	
	SEALEVEL	23.25	23.25	23.75	24.50	25.25	26.25			154	151	131	37.0	
67 PERCENT	10000	18.75	19.00	19.50						139	159	138	27.5	25.0
175 BHP	8000	19.25	19.50	20.00	20.50	21.25				141	158	136	29.0	25.0
	0009	19.75	20.00	20.50	21.00	21.75	22.75			144	154	134	29.0	25.0
	4000	20.25	20.50	21.00	21.50	22.25	23.25	24.25		146	152	132	30.0	25.0
	2000	20.75	21.00	21.50	22.00	22.75	23.75	24.75		149	150	130	31.0	25.0
	SEALEVEL	21.25	21.50	22.00	22.50	23.25	24.25	25.25		151	148	129	32.5	25.0
59 PERCENT	14000	16.00	16.25							126	154	134	23.0	23.0
152.5 BHP	12000	16.50	16.75	17.00	17.50					128	152	133	24.0	23.0
	10000	17.00	17.25	17.50	18.00	18.75	19.50			130	150	131	24.5	23.0
	8000	17.50	17.75	18.00	18.50	19.25	20.00	20.75		134	149	130	25.0	23.0
	0009	18.00	18.25	18.50	19.00	19.75	20.50	21.25	22.25	137	147	128	25.0	23.0
	4000	18.50	18.75	19.00	19.50	20.25	21.00	21.75	22.75	139	145	126	26.0	23.0
	2000	19.00	19.25	19.50	20.00	20.75	21.50	22.25	23.25	141	143	124	26.5	23.0
	SEALEVEL	19.50	19.75	20.00	20.50	21.25	22.00	22.75	23.75	143	141	122	27.5	23.0

DPTIMUM RPM	2400	2400 75 PERCENT	
	2200	2200 67 PERCENT	
	2000	2000 59 PERCENT	

Landing Performance

Landing Distance

LANDING DISTA	ANCE IN FT AT ISA +30	ОС		
	LANDING WEIGHT			
ALTITUDE	4000	5000	6000	6600
0	1300	1400	1480	1590
1000	1340	1450	1550	1640
2000	1410	1500	1590	1700
3000	1470	1560	1660	1760
4000	1540	1620	1720	1830
5000	1580	1680	1770	1880
6000	1650	1740	1840	1940
7000	1700	1800	1900	2000

^{*}SUBTRACT 50FT FOR EVERY 10 C LOWER

Safe Landing Speed

SAFE LANDING	SPEED
WEIGHT	KNOTS
4000	51
4500	53
5000	54
5500	56
6000	57
6500	59
7000	60

^{**}A 5 KNOT TAIL WIND MAY INCREASE LANDING DISTANCE BY UP TO 250 FT WIND AND AIRFIELD SLOPE ARE NOT CONSIDERED IN THIS CHART